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BARREL ASSEMBLY FOR FIREARMS

CROSS REFERENCE TO RELATED APPLICATIONS

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5 This application is a continuation-in-part of application Ser. No. 09/670,162 filed September 26, 2000 now abandoned, which is a divisional of application Ser. No. 09/124,574 filed July 29, 1998, now U.S. 6,123,007 issued September 26, 2000, which itself is a divisional of application Ser. No. 08/525,705 filed as application No. PCT/AU94/00124 on March 14, 1994, now U.S. 5,883,329 issued March 16, 1999.

10 BACKGROUND OF THE INVENTION

The invention relates to firearms.

15 The invention has utility as an automatic, high rate of fire, firearm whereby it may be used for example, as a close-in ship-board defense against bombs, missiles or attack aircraft for launching large numbers of projectiles within a short period of time. The invention also has utility in hand guns such as a rapid fire pistol or rifle which may be disposable.

20 Currently, most firearms use cartridge ammunition which is mechanically fed to a barrel. Such firearms have numerous moving parts, tend to be heavy and complex, may jamb or be unreliable, and require elaborate delivery and loading systems to support the rate of fire. The rate of fire of automatic firearms of this type is limited by the time required to load the cartridge, seal the barrel, unseal the barrel and eject the empty case.

25 More recently, firearms have begun to utilise caseless ammunition which obviates the need to eject an empty case subsequent to firing. However, these firearms retain many of the problems of conventional firearms.

30 SUMMARY OF THE INVENTION

The present invention aims to provide an alternative barrel assembly for a firearms system which will alleviate at least one of the disadvantages of the prior art.

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5 According to one aspect this invention provides a barrel assembly for firearms, said barrel assembly including:

a plurality of projectile assemblies axially disposed in end to end abutting

discrete propellant charges accommodated within said cylindrical spacer portion for propelling respective projectile assemblies sequentially through the

ignition means for igniting said discrete propellant charges; and

The rearward end of the cylindrical spacer portion is preferably adapted to abut the forward or leading end of the subsequent projectile assembly. The cylindrical spacer portion may be expanded into operative sealing contact with said bore of the barrel. Suitably, the interior of the cylindrical spacer portion is structurally reinforced to prevent excessive radial expansion. Desirably, the trailing cylindrical extension of the projectile extends in close proximity with the barrel.

a barrel having a muzzle;

an internal wedging surface, at or adjacent the trailing end of said cylindrical extension which accommodates a tapered nose part of the following projectile, for

discrete propellant charges for propelling respective projectiles sequentially through the muzzle of said barrel;

The trailing cylindrical extension may at least partly define a propellant space therein. Alternatively the propellant charges may surround the noses of respective following projectiles externally of the trailing cylindrical extension.

20 The spacer may include support for the trailing cylindrical extension which may be a thin cylindrical rear extension of the projectile head. The spacer may be integral with the head and trailing cylindrical extension or it may be formed separately therefrom and from a different material if required.

25 In such arrangements the projectile head and spacer may be loaded into the barrel and thereafter an axial displacement thereof caused to expand the trailing end and enhance the sealing engagement between the projectiles and the barrel. This axial displacement is suitably caused individually as the projectiles are subsequently
30 loaded into the barrel. The radial expansion into enhanced sealing engagement with the barrel may be limited through engagement between the penetrating nose of a following projectile and the internal spacer assembly.

The ignition means may be electrical, chemical, mechanical or any other conventional primer. Conveniently, the ignition means is electrical and the control means is an electrical control adapted to provide electrical ignition pulse to the respective ignition means. Suitably the control means is configured to enable a user to selectively control the rate, number, and frequency of the pulses to provide a desired firing pattern. The control means may fire the projectile assemblies singly, in pairs, or in any other combinations.

The projectile assembly may be round, conventionally shaped or dart-like and the fins thereof may be off-set to generate a stabilising spin as the dart is propelled from a barrel which may be a smooth-bored barrel. In addition the barrel assembly may find utility as a removable/replaceable barrel of a rifle or pistol.

Alternatively the barrel assembly constitutes one of a plurality of barrel assemblies and the control means may actuate the ignition means of each of the barrel assemblies in such manner that a sequential plurality of arrays of projectile assemblies are propelled in following relationship. Aiming and firing of the arrays of projectile assemblies may be controlled by a conventional radar fire control system or other known fire control systems. The individual barrel assemblies may be aimed such that the array of projectile assemblies converges at a particular range to give a maximum density of projectile assemblies at that range.

Alternatively, the array of projectile assemblies may diverge to maximise coverage of an area. Thus, the average separation distance at the target between the projectile assemblies in an array can be predetermined and adjusted to suit the nature and range of the target. Of course, the individual barrel assemblies may be fired randomly or independently of the other barrel assemblies.

The propelling charges may be either solid or granular and compression of either may be an undesirable, moreover, movement of the projectile assemblies relative to the barrel may cause misalignment of the ignition means with their respective propellant charges.

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It is preferred that the ignition means be disposed at the leading end of the propellant charge so as to minimise possible energy loss in accelerating the front portion of the propellant charge.

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In another embodiment, complementary wedging surfaces are disposed on the spacer assembly and projectile head respectively whereby the projectile head is urged into engagement with the bore of the barrel in response to relative axial compression between the spacer means and the projectile head. In such arrangement the projectile head and spacer assembly may be loaded into the barrel and thereafter an axial displacement is caused to ensure good sealing between the projectile head and barrel. Suitably the extension means is urged into engagement with the bore of the barrel.

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Preferably, the projectile head defines a tapered aperture at its rearward end into which is received a complementary tapered spigot disposed on the leading end of the spacer assembly, wherein relative axial movement between the projectile head and the complementary tapered spigot causes a radially expanding force to be applied to the projectile head.

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The barrel may be non-metallic and the bore of the barrel may include recesses which may fully or partly accommodate the ignition means. In this situation the barrel houses electrical conductors which facilitate electrical communication between the control means and ignition means. This arrangement may be utilised for disposable barrel assemblies which have a limited firing life and the ignition means and control wire or wires therefor can be integrally manufactured with the barrel.

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In an alternative arrangement, a barrel assembly includes ignition apertures in the barrel and the ignition means are disposed outside the barrel and adjacent the apertures. The barrel may be surrounded by a non-metallic outer barrel which may include recesses adapted to accommodate the ignition means. The outer barrel may also house electrical conductors which facilitate electrical communication

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between the control means and ignition means. The outer barrel may be formed as a laminated plastics barrel which may include a printed circuit laminate for the ignition means.

- 5 Both of the above arrangements lend themselves to a modular or disposable construction. The barrel assemblies may be adapted for firing as is, or may be adapted for mounting within a housing.

- 10 For safety, the barrel assembly may include an arming switch associated with each ignition means which is closed in response to the preceding projectile assembly being discharged. Preferably, the arming switch is closed by biasing means which are normally resisted by the preceding projectile assembly. In a preferred embodiment, the projectile head and spacer assembly each constitute switch contacts which are normally electrically isolated from each other and wherein an
15 electrical circuit between the barrel and spacer body is completed in response to the preceding projectile assembly being discharged. In this arrangement, the barrel, which is in electrical contact with the projectile head, is also in contact with one of the electrodes.

20 BRIEF DESCRIPTION OF THE DRAWINGS

In order that this invention may be more readily understood and put into practical effect, reference will now be made to the accompanying drawings which illustrate typical embodiments of the invention and wherein:

- 25 FIG. 1 is a sectional and schematic view of an embodiment of a barrel assembly according to the invention;

FIG. 2 schematically illustrates the concept of a plurality of barrel assemblies according to the invention being massed in pods;

- 30 FIG. 3 is a schematic view of arrays of projectile assemblies being fired from the pods of FIG. 2;

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FIGS. 14 and 15 are sectional side and end views of another embodiment of a projectile of the type generally illustrated in FIG. 5;

FIG. 16 illustrates a barrel assembly employing projectiles of the type illustrated in FIGS. 14 and 15; and

5 FIG. 17 illustrates a barrel assembly of a further embodiment of the invention.

DESCRIPTION OF THE PREFERRED EMBODIMENTS

Referring to FIG. 1, there is illustrated a barrel assembly 10 including a barrel 12, a plurality of projectile assemblies 14, here of simple spherical form, axially disposed within barrel 12 for operative, sealing engagement with the bore of barrel 12, discrete propellant charges 16 disposed between adjacent projectile assemblies 14 for propelling the respective projectile assemblies 14 individually and sequentially through the muzzle of barrel 12, ignition means 18 for igniting discrete propellant charges 16, and control means 20 for selectively and sequentially actuating ignition means 18.

In use, the leading projectile assembly 14 is propelled in response to ignition of the leading propellant charge 16 by the leading ignition means 18. Thereafter the following projectile assemblies are sequentially propelled in like fashion. There is no ammunition delivery system or moving parts, and the firing rate is practically limited only by the time taken for each projectile assembly to exit the barrel.

The control means may have time delay means to control the rapidity of fire and or timing means permitting a selected number of sequential ignitions in response to each manual actuation of the ignition means, such as by squeezing a trigger. A mode switch may be associated with the control means to enable a user to select the form of firing, ie full barrel discharge, short bursts of rapid fire, sequential fire of a selected number of projectiles, single shot firing per actuation etc. Integrated circuit electronic control means are preferably utilised as the control means and may be manufactured as part of the barrel assembly.

Referring to FIG. 2, the barrel assembly constitutes one of a plurality of barrel assemblies and the control means actuates the ignition means of each of the barrel

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assemblies in such manner that a sequential plurality of arrays of projectile assemblies are propelled in following relationship as shown in FIG. 3. The plurality of barrel assemblies forms a pod 22 and a plurality of pods are mounted on a trainable mount 24. The aiming and firing of the barrel assemblies is controlled by a radar fire control system 25 or other conventional system.

In one form, each barrel is 2.25 metres long and has an outside diameter of 20 mm. The combined propelling charge/projectile assembly length is 50 mm. Leaving 0.25 metres of the barrel free, 40 projectile assemblies together with their associated propellant charges can be pre-loaded into the barrel. The pod has a cross-sectional dimension of 0.75 metres by 0.75 metres for example and therefore accommodates approximately 1200 barrel assemblies. Thus, a pod can be pre-loaded with 48000 projectile assemblies.

This enables significant fire-power to be associated with a relatively small weapon and a very high discharge rate to be achieved, bearing in mind the firing rate of each individual barrel assembly may be significantly in excess of the rate achievable by conventional automatic firearms. The barrel assemblies may be formed as a relatively lightweight honeycomb structure which will be very stiff and if desired the barrels may be arranged to focus at a point relatively close to the weapon with a view to counteracting the spreading tendencies produced by the expansion of the hot explosion gases radiating in an outwards direction. Alternatively a box-like baffle could be used to prevent the immediate outward spread of the gases. This baffle may be slidably supported about the outer barrel section for extension past the end of the barrels during firing. A further manner of alleviating this perceived effect would be to slightly stagger the firing of the projectiles.

Referring to the embodiments of FIGS. 4 to 10, projectile assemblies 14 are disposed in axial abutting relationship to form a compression resistant column. Axially compressive loads are created by the pressures generated in the barrel by the propulsion of preceding projectile assemblies. Compression can result in an

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alteration of the burn rate of a propelling charge, misalignment of ignition means with respective propelling charges or even premature ignition of propelling charge.

Each projectile assembly 14 includes a projectile head 26 and means for defining a propellant space in the form of spacer assembly 28 which extends axially and rearwardly from projectile head 26 and abuts an adjacent projectile assembly 14.

In one form, the projectile head 26 is formed from a heavy malleable material such as lead to facilitate operative sealing with barrel 12, and spacer assembly 28 is formed of a rigid material such as steel.

In the embodiment of FIG. 5, the spacer assembly 28 takes the form of a cylinder axially extending from projectile head 26. The interior of the cylinder accommodates propellant charge 16 and is structurally reinforced 27 to prevent excessive radial expansion. The end 29 of the cylinder is adapted to abut the leading end of the subsequent projectile assembly 14. Further embodiments employing projectile assemblies formed with an integral cylindrical extension are described below in relation to FIGS. 14 to 17

Referring to the embodiments of FIGS. 6 and 7, spacer assembly 28 extends through projectile head 26 to the leading end of projectile head 26 whereby compressive loads are transmitted directly between adjacent spacer assemblies 28. Spacer assembly 28 supports a thin cylindrical rear portion 30 of projectile head 26 in operative sealing contact with the bore of barrel 12. Specifically, spacer assembly 28 includes a radially outwardly extending collar flange 32 which supports thin cylindrical rear portion 30 of projectile head 26 in operative sealing contact with the bore of barrel 12.

Referring to the embodiments of FIGS. 9 and 10, complementary wedging surfaces 34, 36 are disposed on spacer assembly 28 and projectile head 26 respectively whereby thin cylindrical rear portion 30 of projectile head 26 is urged into engagement with the bore of barrel 12 in response to an axially compressive load being applied to projectile assembly 14. Projectile head 26 defines a tapered

aperture 38 at its rearward end into which is received a complementary tapered spigot 40 disposed on the leading end of spacer assembly 28. Relative axial movement between tapered aperture 38 and complementary tapered spigot 40 causes a radially expanding force to be applied to thin cylindrical rear portion 30 of projectile head 26.

In the embodiment of FIG. 7, barrel 12 is non-metallic and the bore of the barrel includes recesses 42 which at least partly accommodate ignition means 18. Barrel 12 may be formed of kevlar, carbon fibre, glass reinforced polymer or the like. Thus, the barrel assembly may be lightweight and disposable. Barrel 12 houses electrical conductors 44 which facilitate electrical communication between the control means and ignition means.

In the embodiments of FIGS. 8 and 9, barrel 12 includes ignition apertures 46 and ignition means 18 are disposed outside the barrel and adjacent the apertures. Barrel 12 is surrounded by a non-metallic outer barrel 48, the bore of the outer barrel including recesses adapted to at least partly accommodate the ignition means. The barrel assembly may be slidably received in sheath 50. Outer barrel 48 houses electrical conductors 44 which facilitate electrical communication between the control means and ignition means 18.

Referring to FIG. 10, arming switch 52 associated with ignition means 18 is closed in response to the preceding projectile assembly being discharged. Specifically, arming switch is closed by biasing means 54 once the preceding projectile assembly has been propelled. Projectile head 26 and spacer assembly 28 each constitute switch contacts which are normally electrically isolated from each other by insulating layer 56. An electrical circuit between barrel 12 and spacer assembly 28 is completed when arming switch 52 closes in response to the preceding projectile assembly being discharged. The ignition means 18 is thus armed only when the preceding projectile assembly has been discharged.

A four barrel hand gun 60 is illustrated in FIG. 11. The barrels of the four barrel set 61, are arranged in a square formation, and are fed by a matching replaceable four

5 The four barrel magazine block 62 is loaded with 5 rounds per barrel, which number may of course be varied depending on the size of the block and the size of the round. In this embodiment the magazine block 62 contains twenty rounds.

The projectiles for use with the above described embodiments may be provided with external flights or spiral ridges as illustrated in FIGS. 12 and 13. The ridges 70 are provided on the nose of the projectile to impart spin during flight. In the form illustrated a 7.62 mm bullet 71 has four spiralling ridges 70 radiating from the nose of the bullet. The ridges are of an average height of 1.5 mm and extend the length of the nose of the bullet, but not along the side of the bullet. The pitch is suitably formed as to provide a single revolution of the bullet about its longitudinal axis for every meter travelled.

Of course two or more spiralling ridges, spaced evenly around the bullet nose may be utilised if desired. Furthermore the height of the ridges, the length of the ridges, the pitch or degree of spiralling, the geometric curve form of the spiral, may be varied to suit the desired flight characteristics. The ridges may also extend along the side of the bullet. The cross section profile of the spiral ridges may be relatively flat,

or steep according to the intended use of the ammunition, and the desired degree of reaction to the airflow.

As illustrated in FIG. 13, the ridges 70 may have a steep leading face 72, which offers resistance to the airflow over the bullet, and causes the bullet to rotate, a flat top portion 73 and trailing faces 74 which slope gently to the surface of the bullet.

Such ammunition may also be used in rifled barrel weapons to advantage. Also as the spirals on the bullet would assist in producing the spin during firing, the normal pressure applied by the edge of the rifling lands against the soft metal of the bullet would be reduced. Therefore the bullet would not require the rifling to cut as long a track along the side of the bullet. Rather, the small expanding band of the Minie gas sealing system would then be adequate to assist with spin acceleration. On impact with soft targets, the spiral bullet of the present invention would tend to react to the increased pressure on the ridges by maintaining a high rate of twist, as it progresses through the target material.

The projectile 14 illustrated in FIGs 14 and 15 includes a thin cylindrical extension 30 integral with the projectile head 26 and extending axially therefrom in close conformity with the bore of the barrel 12, as shown in the barrel assembly 10 in FIG. 16. Internal reinforcement is provided for the trailing cylindrical extension 30. The reinforcement includes webs 27 extending radially from a central spine 28 as seen in cross-section in FIG. 15, which spine extends axially from the head 26. This reinforcement prevents excessive radial expansion of the trailing cylindrical extension 30 occurring during firing.

The trailing end of the projectile 14 has a central recess 31 formed in the end of the spine 28 and associated radial webs 27 to accommodate the nose 25 of a following projectile 14b. The projectile head 26 and associated spine 28 enables a continuous compression resistant column to be formed when the projectiles 14 are stacked in abutting relationship in the barrel 12.

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In the barrel assembly 10 illustrated in FIG 16, a leading projectile 14a is shown slightly separated from an adjacent following projectile 14b and partly advanced through the barrel 11, while the rearmost projectile 14c nest in end to end abutting relationship to form an axial stack of projectiles 14 within the barrel 12.

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The cylindrical extension 30 also accommodates the propellant charge 16 which is selectively ignitable by an electronically controlled ignition means 18, which ignition means may be internal or external of the barrel. The radial webs 27 are streamlined to assist gas flow thereabout for discharge through the open rear end 29 of the cylindrical extension 30.

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An axial stack of projectiles 80 of a further embodiment of the invention as illustrated in FIG 17, are similar to the projectiles 14 of FIG 16. However in this form, the central spine 81 which extends rearward from the projectile head 85 terminates short of the rearmost radial webs 82. The rearmost webs are profiled to closely accommodate the rearwardly diverging nose portion 83 of a following projectile. Further radial webs 87 extend from the spine 81 and provide support for the intermediate portion of the thin cylindrical extension 88.

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In one arrangement, a leading projectile 84a may be set into position in a supporting barrel (not illustrated) by forcing its rearmost webs 82 over the nose portion 83 of a following projectile 84b. This forcing action causes a slight radial expansion (shown in exaggerated form) of the trailing end 86 of the projectile and assists the projectiles 84 to seal in a supporting barrel.

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The expansion may be limited by terminating the spine 81 a set distance forward of the rearmost webs 82 to form a stop for the penetrating nose 83 of the following projectile 84b. Alternatively the amount of penetration may be set by applying a predetermined impact to the projectile being loaded to achieve the desired radial engagement of the trailing end 86 of the leading projectile 84a with the supporting barrel (not shown). Suitably this radial engagement is achieved before the nose 83 of the following projectile abuts the spine 81, as denoted by 89.

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The radial engagement may be increased in use due to gas pressure applied to a following projectile. The gas pressure is generated upon firing the propellant charge associated with the leading projectile. That is, the trailing end 86 of a leading projectile may be forced further over of the nose portion 83 of its respective
5 following projectile until the leading spine 81 abuts the following nose 83.

A rear collar or thickening of the cylindrical extension 88 may be added to or substituted for the radial webs 87 if desired, or the configuration of four radial webs (as illustrated) may be replaced by numerous webs closely spaced about the inner
10 periphery of the trailing end of the cylindrical portion 88. These variations being provided in order to achieve a substantially uniform enhanced sealing engagement of the trailing end 86 of projectiles 84 with the barrel 12.

In the embodiments illustrated in FIGS 16 and 17, sealing of the projectile in the
15 barrel is at least partially effected by the relatively long cylindrical extension 30, 88 being closely accommodated within the bore of the supporting barrel 11. Additionally, in the case of the FIG 17 embodiment, sealing is further assisted by the radial expansion of the trailing end 86 into enhanced sealing engagement with the supporting barrel. Furthermore the projectiles 14, 84 may each carry their own
20 supply of propellant 16 to facilitate convenient loading of a barrel 12. Extra propellant may be accommodated in the space between and about the nose of the projectiles if desired.

It will of course be realised that the above has been given only by way of illustrative
25 example of the invention, and that all such modifications and variations thereto as would be apparent to persons skilled in the art are deemed to fall within the broad ambit and scope of the invention as is herein set forth in the accompanying claims.

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